

Lecture 13

Advanced Vibrations

TODAY

Solutions To Midterm

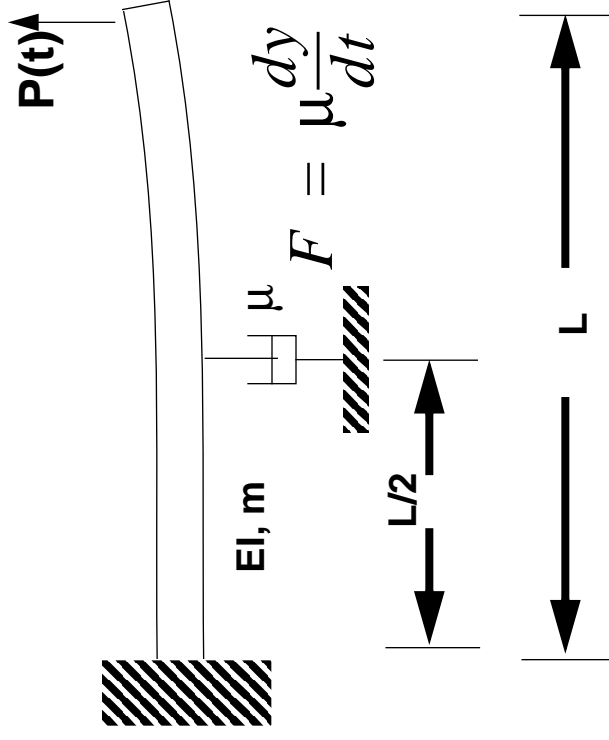
Eigen Solutions to the Differential Equations Damping Matrices Modal Damping

Mid-Term Take Home Exam

Problem 1. Assumed Modes

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Consider the cantilevered Beam shown here. There is a time-dependent load applied at the end and a dashpot placed at the center of the span.



Assume a single deformation mode:

$$y(x, t) = A_1(t) \left(\frac{x}{L} \right)^2.$$

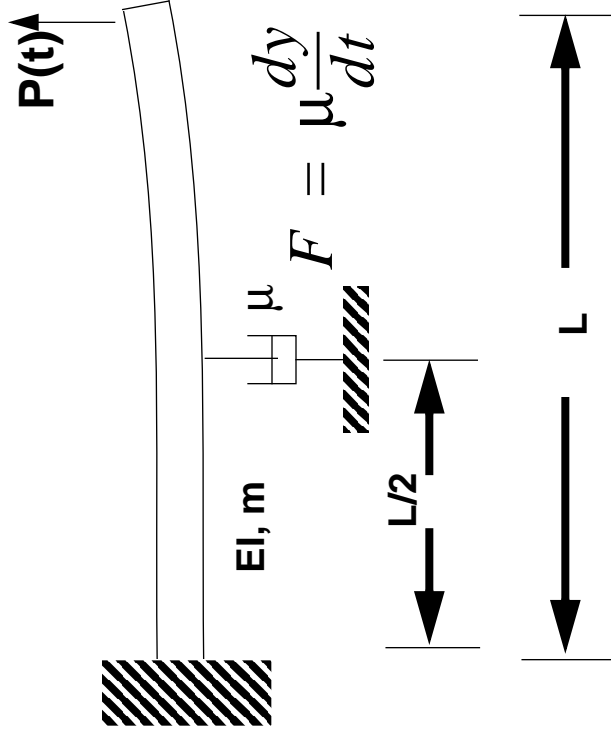
- Derive the governing equation for $A_1(t)$.
- Assuming that P is harmonic, calculate the complex magnification factor for the tip displacement as a function of the frequency.
- Calculate the time-averaged rate of energy dissipation as a function of frequency and force amplitude.

Solution to Problem

$$y(x, t) = A_1(t) \left(\frac{x}{L} \right)^2.$$

$$T = \int_0^L \frac{m}{2} A^2 \left(\frac{x}{L} \right)^4 dx = \frac{Lm}{2 \cdot 5} \dot{A}^2$$

$$V = \int_0^L \frac{EI}{2} A^2 \left(\frac{2}{L^2} \right)^2 dx = \frac{4EI}{2L^3} A^2$$



Solution to Problem

Applied force: $\delta W = Q_P \delta A = P \delta y(L) \Rightarrow Q_P = P$

Damping force:

$$\delta W = Q_D \delta A = F_D \delta y\left(\frac{L}{2}\right) = -\mu \left(\frac{\dot{A}}{4}\right) \frac{\delta A}{4} \Rightarrow Q_D = -\frac{\mu \dot{A}}{16}$$

The Lagrange equation is $\frac{Lm}{5} \ddot{A} + \frac{\mu \dot{A}}{16} + \frac{4EI}{L^3} A = P(t)$

In more familiar terms, this is $M \ddot{A} + C \dot{A} + KA = P$

where $M = \frac{Lm}{5}$, $C = \frac{\mu}{16}$, and $K = \frac{4EI}{L^3}$

Solution to Problem

Assume that $P(t) = \operatorname{Re}\{P_0 e^{i\omega t}\}$, then we expect the resulting displacement will be $A(t) = \operatorname{Re}\{A_0 e^{i\omega t}\}$. Substitution into the governing equation yields

$$\operatorname{Re}\{(-\omega^2 M + i\omega C + K)A_0 - P_0\}e^{i\omega t} = 0$$

from which we conclude that $[-\omega^2 M + i\omega C + K]A_0 = P_0$

$$\text{which is re-arranged to } \left[1 - \left(\frac{\omega}{\omega_0}\right)^2 + 2i\zeta \frac{\omega}{\omega_0}\right]A_0 = \frac{P_0}{K}$$

$$\text{where } \omega_0^2 = \frac{K}{M} = \frac{20EI}{mL^4} \text{ and } \zeta = \frac{C}{2M\omega_0} = \frac{\mu L}{64} \sqrt{\frac{5}{mEI}}$$

Solution to Problem

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Solving for A_0 in terms of P_0 ,

$$A_0 = \frac{P_0}{K} H(\omega)$$

where $H(\omega) = \frac{1}{\left[1 - \left(\frac{\omega}{\omega_0}\right)^2 + 2i\zeta\frac{\omega}{\omega_0}\right]}$ is the Magnification Factor.

Solution to Problem

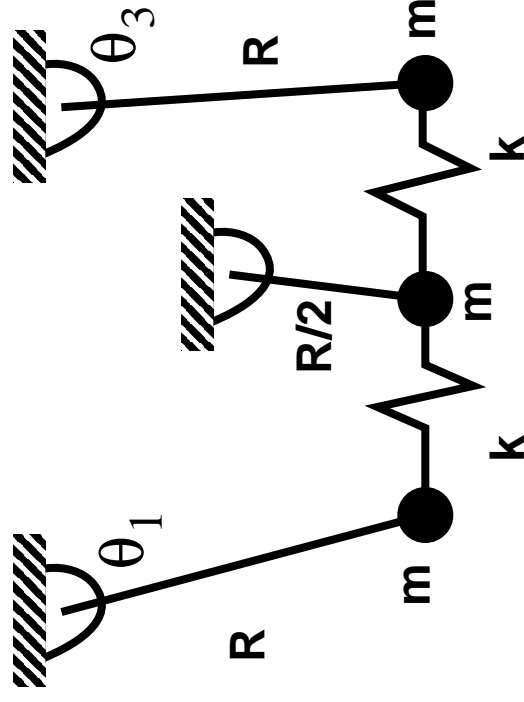
The average rate of energy dissipation is

$$\begin{aligned} D &= \frac{\omega}{2\pi} \int_0^{2\pi/\omega} P(t) \dot{y}(L, t) dt \\ &= \frac{\omega}{2\pi} \int_0^{2\pi/\omega} \operatorname{Re}\{P_0 e^{i\omega t}\} \operatorname{Re}\{i\omega A_0 e^{i\omega t}\} dt \\ &= \frac{1}{2} \operatorname{Re}\{(P_0^*)(i\omega A_0)\} = \frac{1}{2} \operatorname{Re}\{(P_0^*)(i\omega P_0 H(\omega)/K)\} \\ &= \omega \frac{L^3 |P_0|^2}{8EI} \operatorname{Re}\{iH(\omega)\} = -\omega \frac{L^3 |P_0|^2}{8EI} \operatorname{Im}\{H(\omega)\} \end{aligned}$$

Problem 2.

Consider the three pendula shown.

- Calculate the governing equations for the three angles θ_1 , θ_2 , & θ_3 in terms of the parameters shown. (Ignore vertical components of spring extension.)
- Linearize the equations, evaluating the mass and stiffness matrices.



- Impose the constraint that $\theta_1 = \theta_3$, writing the constraint matrix. Write the mass and stiffness matrices in the reduced system.
- Calculate the eigenmodes and frequencies in the reduced system.
- Express those eigenmodes in terms of the full set of displacement degrees of freedom, θ_1 , θ_2 , & θ_3 .

Solution

Derivation of governing equations:

$$V = \frac{k}{2} \left[\left(\frac{R}{2} \right) \sin \theta_2 - R \sin \theta_1 \right]^2 + \frac{k}{2} \left[\left(\frac{R}{2} \right) \sin \theta_2 - R \sin \theta_3 \right]^2 - mgR \cos \theta_1 - mg \frac{R}{2} \cos \theta_2 - mgR \cos \theta_3 .$$

$$\frac{\partial V}{\partial \theta_1} = mgR \theta_1 + kR^2 \left(\theta_1 - \frac{\theta_2}{2} \right)$$

$$\frac{\partial V}{\partial \theta_2} = mg \frac{R}{2} \theta_2 + \frac{kR^2}{2} (\theta_2 - \theta_1 - \theta_3) \text{ and}$$

$$\frac{\partial V}{\partial \theta_3} = mgR \theta_3 + kR^2 \left(\theta_3 - \frac{\theta_2}{2} \right) \text{ after linearization.}$$

Solution

$$T = \frac{m}{2} R^2 \dot{\theta}_1^2 + \frac{m}{2} \left(\frac{R}{2} \right)^2 \dot{\theta}_2^2 + \frac{m}{2} R^2 \dot{\theta}_3^2$$

Yielding the governing equations

$$\frac{R}{g} \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1/4 & 0 \\ 0 & 1 & 1 \end{bmatrix} \begin{bmatrix} \ddot{\theta}_1 \\ \ddot{\theta}_2 \\ \ddot{\theta}_3 \end{bmatrix} + \begin{bmatrix} (1+\gamma) & -\gamma/2 & 0 \\ -\gamma/2 & (1+\gamma)/2 & -\gamma/2 \\ 0 & -\gamma/2 & (1+\gamma) \end{bmatrix} \begin{bmatrix} \theta_1 \\ \theta_2 \\ \theta_3 \end{bmatrix} = 0$$

where $\gamma = (kR)/(mg)$

Solution

$$\text{Setting } \begin{bmatrix} \theta_1 & \theta_2 & \theta_3 \end{bmatrix}^T = \text{Re} \left\{ e^{i\omega t} \begin{bmatrix} A_1 & A_2 & A_3 \end{bmatrix}^T \right\}$$

The resulting algebraic eigen problem becomes

$$-\alpha^2 \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1/4 & 0 \\ 0 & 1 & 1 \end{bmatrix} \begin{bmatrix} A_1 \\ A_2 \\ A_3 \end{bmatrix} + \begin{bmatrix} (1+\gamma) & -\gamma/2 & 0 \\ -\gamma/2 & (1+\gamma)/2 & -\gamma/2 \\ 0 & -\gamma/2 & (1+\gamma) \end{bmatrix} \begin{bmatrix} A_1 \\ A_2 \\ A_3 \end{bmatrix} = 0$$

$$\text{where } \alpha^2 = R\omega^2/g$$

Solution

The constraint that $\theta_1 = \theta_3$ yields the constraint equation

$$\begin{bmatrix} A_1 \\ A_2 \\ A_3 \end{bmatrix} = T \begin{bmatrix} A_1 \\ A_2 \end{bmatrix} \text{ where } T = \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ 1 & 0 \end{bmatrix}. \text{ The new matrices are}$$

$$\bar{\bar{K}} = T^T K T = \begin{bmatrix} 2(1 + \gamma) & -\gamma \\ -\gamma & \frac{1}{2}(1 + \gamma) \end{bmatrix} \text{ and}$$

$$\bar{\bar{M}} = T^T M T = \begin{bmatrix} 2 & 0 \\ 0 & 1/4 \end{bmatrix}$$

Solution

The eigen problem in this reduced system is

$$-\alpha^2 \begin{bmatrix} 2 & 0 \\ 0 & 1/4 \end{bmatrix} \begin{bmatrix} A_1 \\ A_2 \end{bmatrix} + \begin{bmatrix} 2(1+\gamma) & -\gamma \\ -\gamma & \frac{1}{2}(1+\gamma) \end{bmatrix} \begin{bmatrix} A_1 \\ A_2 \end{bmatrix} = 0.$$

$$\det \begin{bmatrix} 2(1+\gamma) - 2\alpha^2 & -\gamma \\ -\gamma & \frac{1}{2}(1+\gamma) - \frac{\alpha^2}{4} \end{bmatrix} = 0$$

This has a characteristic equation $\alpha^4 - \frac{3}{2}(1+\gamma)\alpha^2 + (1+2\gamma) = 0$

Solution

The solutions are $\alpha^2 = \frac{3}{2}(1 + \gamma) \pm \frac{1}{2}\sqrt{1 + 2\gamma + 9\gamma^2}$. Substituting this back into the algebraic eigen value problem

$$\begin{bmatrix} -(1 + \gamma) \mp \sqrt{1 + 2\gamma + 9\gamma^2} & -\gamma \\ -\gamma & \frac{1}{8}[(1 + \gamma) \mp \sqrt{1 + 2\gamma + 9\gamma^2}] \end{bmatrix} \begin{bmatrix} 1 \\ \frac{A_2}{A_1} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$$

$$\text{yields } \frac{A_2}{A_1} = -\frac{1 + \gamma}{\gamma} \mp \frac{\sqrt{1 + 2\gamma + 9\gamma^2}}{\gamma}$$

Solution

Map these modes back to original space

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$$\begin{bmatrix} A_1 \\ A_2 \\ A_3 \end{bmatrix} = T \begin{bmatrix} A_1 \\ A_2 \end{bmatrix} \text{ where } T = \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ 1 & 0 \end{bmatrix}, \text{ so our solutions are}$$

$$\left(\alpha^2 = \frac{3}{2}(1 + \gamma) \pm \frac{1}{2}\sqrt{1 + 2\gamma + 9\gamma^2} \right), \begin{bmatrix} 1 & \\ -\frac{1 + \gamma}{\gamma} \pm \frac{\sqrt{1 + 2\gamma + 9\gamma^2}}{\gamma} & 1 \end{bmatrix}$$

Examine Solution

Consider case where gravity dominate over spring forces: $\gamma = 0$. The two solutions are $\alpha^2 = \frac{3}{2} \pm \frac{1}{2}$:

$$\alpha^2 = 1 \text{ corresponds to } \begin{bmatrix} A_1 \\ A_2 \end{bmatrix} = \begin{bmatrix} 1 \\ 0 \end{bmatrix} \text{ and}$$

$$\alpha^2 = 2 \text{ corresponds to } \begin{bmatrix} A_1 \\ A_2 \end{bmatrix} = \begin{bmatrix} 0 \\ 1 \end{bmatrix}$$

as expected.

Examine Solution

Consider case where spring forces dominate over gravity: $\gamma \rightarrow \infty$. The

two solutions are $\alpha^2 = \left(\frac{3}{2} \pm \frac{3}{2}\right)\gamma$:

$\alpha^2 = 0$ corresponds to $\begin{bmatrix} A_1 \\ A_2 \end{bmatrix} = \begin{bmatrix} 1 \\ 2 \end{bmatrix}$ (rigid body motion), and

$\alpha^2 = 3\gamma$ corresponds to $\begin{bmatrix} A_1 \\ A_2 \end{bmatrix} = \begin{bmatrix} 1 \\ -4 \end{bmatrix}$

as expected.

Solutions to Governing Equations

Lets consider the solution $x(t)$ to $M\ddot{x}(t) + Kx(t) = F(t)$ and

expand $x(t) = \sum_{k=1}^N \beta_k(t)x_k$ where x_k is the k 'th normalized

eigenvector, corresponding to the k 'th eigen value ω_k^2 . This can be expressed in matrix form as $x(t) = P\beta(t)$ where P is modal matrix.

This yields the system $\ddot{\beta} + D\beta = P^T F = \mathcal{F}$ where $D = P^T K P$ is the diagonal matrix of eigen values.

Solutions to Governing Equations

Lets solve for the β via Laplace Transforms:

$$(s^2 + \omega_k^2) \mathcal{L}\beta_k = s\beta_k(0) + \dot{\beta}_k(0) + \mathcal{L}(\mathcal{F}_k(t))(s)$$

This has the solution

$$\beta_k = \beta_k(0) \cos \omega_k t + \dot{\beta}_k(0) \sin \omega_k t + \int_0^t \mathcal{F}_k(t - \tau) \frac{\sin(\omega_k \tau)}{\omega_k} d\tau$$

where the vector $\dot{\beta}(0) = P^{-1} \dot{x}(0)$ and $\beta(0) = P^{-1} x(0)$

Solutions to Governing Equations With Damping

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The above development provided the solutions for problem with no damping. The more general problem is where the damping is represented by a general matrix C . In general, this will yield complex modes, and we shall study that case later.

There is a class of damping matrices for which the elastic eigen modes are preserved. These problems are much more tractable. Because

$P^T M P = I$ and $P^T K P = D$ we may conclude that

$$M = P^{-T} I P^{-1} \text{ and } K = P^{-T} D P^{-1}.$$

This suggests looking for damping matrices $C = P^{-T} \chi P^{-1}$ where χ is also a diagonal matrix consisting of terms $2\zeta_k \omega_k$ on the diagonal.

Damping Matrices

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We examine our dynamics equation

$$M\ddot{x} + C\dot{x} + Kx = F.$$

We diagonalize with the modal matrix to find

$$\ddot{\beta} + \chi\dot{\beta} + D\beta = P^T F = \mathcal{F}$$

Again, we solve for the β via Laplace Transforms:

$$(s^2 + 2\zeta_k\omega_k s + \omega_k^2)\mathcal{L}\beta_k = s\beta_k(0) + \dot{\beta}_k(0) + \mathcal{F}_k(t) \text{ for each } k.$$

Solution

Exploit that $\mathcal{L}^{-1}\left(\frac{1}{s^2 + 2\zeta_k \omega_k s + \omega_k^2}\right) = e^{-\zeta_k \omega_k t} \frac{\sin(\omega_k \sqrt{1 - \zeta_k^2} t)}{\omega_k \sqrt{1 - \zeta_k^2}}$

and that $\mathcal{L}^{-1}\left(\frac{s}{s^2 + 2\zeta_k \omega_k s + \omega_k^2}\right) = \frac{d}{dt}\left(e^{-\zeta_k \omega_k t} \frac{\sin(\omega_k \sqrt{1 - \zeta_k^2} t)}{\omega_k \sqrt{1 - \zeta_k^2}}\right)$

to solve for $\beta_k(t)$.

Transfer Functions Synthesized from Modal Response

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The whole system is represented in Laplace space by

$$\mathcal{L}(x) = P \mathcal{L}(\beta) = P G^d P^T \mathcal{L}(F)$$

where $G^d(s)$ is a diagonal matrix, whose diagonal terms are

$$G_{kk}^d = \frac{1}{s^2 + 2\zeta_k \omega_k s + \omega_k^2}$$

we have assumed homogeneous initial conditions.

Say we want to find the transfer function from force DOF n to displacement DOF m . Then

$$G_{mn}(s) = \sum_{k=1}^{\text{modes}} P_{mk} G_{kk}^d P_{nk}$$

Modal Damping

The assumption made above, that the damping matrix is diagonalizable, is the assumption of *modal damping*.

Note that modal damping includes the special cases of proportional damping:

$$M\ddot{x} + (aM + bK)\dot{x} + Kx = F(t)$$

is diagonalized by the modal matrix. $x = P\beta$ as

$$\ddot{\beta} + (aI + bD)\dot{\beta} + D\beta = \mathcal{F}.$$

Modal Damping Terms

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We see the advantages of assuming modal damping, the question becomes: “How do we deduce that damping ζ_k associated with each mode?”

There are three standard answers.

1. Experimentally. Plot frequency response, identify peaks with eigen frequencies, and determine fraction for critical damping from the half power points.
2. Where a real damping matrix is known, compute $P^T C P$ and discard the off-diagonal terms
3. Where the damping mechanism can be modeled, set the model damping to reproduce the dissipation ratio associated with modal vibration. This is the modal strain energy method.

Examples and discussion follow.

Next Time

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**More on Modal Damping and the Modal Strain Energy Method.
An Introduction to Linear Viscoelasticity.**